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REPRESENTATIVE OPERATING CHARTS OF PROPELLERS TESTED

IN THE NACA 20-FOOT PROPELLER-RESEARCH TUNNEL

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Langley Field, Va.

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#### ADVANCE RESTRICTED REPORT

#### REPRESENTATIVE OPERATING CHARTS OF PROPELLERS TESTED

IN THE NACA 20-FOOT PROPELLER-RESEARCH TUNNEL

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Extensive tests of full-scale propellers have been made in the 20-foot propeller-research tunnel (PRT) at the Langley Memorial Aeronautical Laboratory during its many years of operation. The results were usually presented in the form of charts showing thrust coefficient, power coefficient, and efficiency, each plotted separately against V/nD. The type of chart now being employed by airplane and propeller manufacturers for performance estimates consists of power coefficients plotted against V/nD with lines of constant efficiency superimposed. As a result of numerous requests to issue this type of chart directly and on a large scale, the present report has been compiled from data taken from a series of fairly recent reports. Charts are presented only for the conditions considered to be useful in present-day design (see table I), although charts covering other conditions have been developed and are available.

In the preparation of the plots presented, the data were cross-faired and do not necessarily check exactly the previously published data. The power coefficients of all dual-rotating propellers in this report represent the sum of the power coefficients of the front and rear propellers and are for the test conditions in which blade angles of the front and rear propellers were set to absorb approximately equal power at peak efficiency only.

The usual coefficients and symbols have been used and are defined as follows:

$$C_{\rm p}$$
 power coefficient  $\left(\frac{\text{engine power}}{\rho_{\rm n} 3_{\rm p} 5}\right)$ 

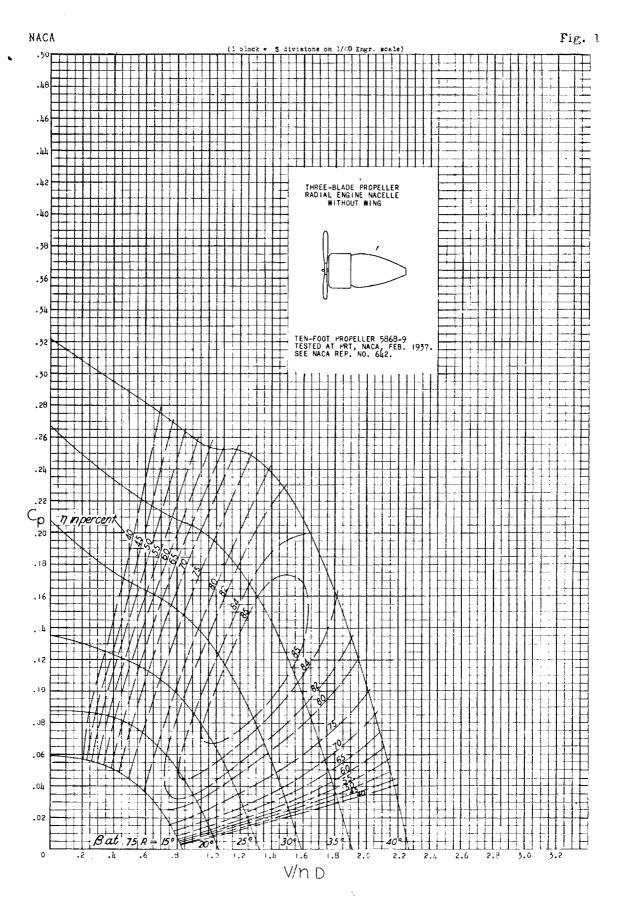
$$c_{\text{T}}$$
 thrust coefficient  $\left(\frac{\text{effective thrust}}{\rho n^2 D^4}\right)$ 

$$\eta$$
 propeller efficiency ( $ext{C}_{ ext{T}}/ ext{C}_{ ext{P}}$ ) ( $ext{V/nD}$ )

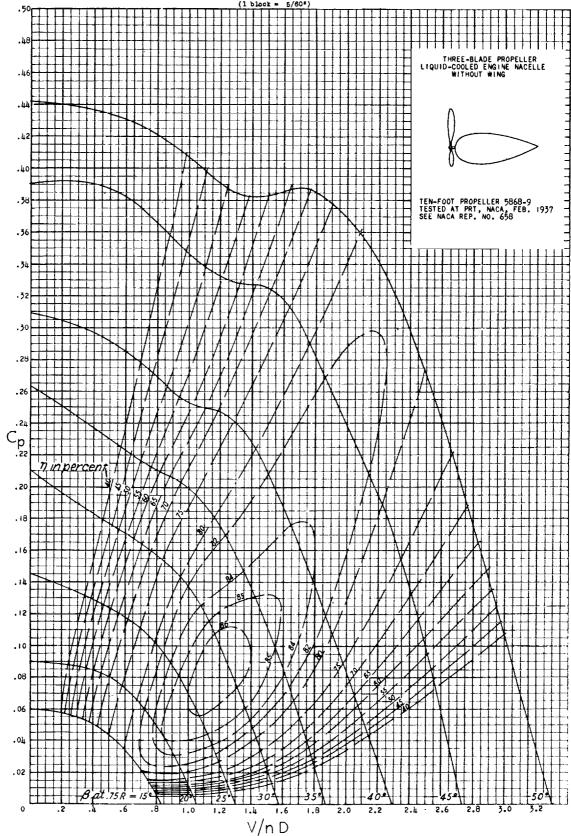
and

mass density of air, slugs per cubic foot

F1gur•	Blade activity factor	Number of blades	Rotation	Propeller location	Body configuration	Reference
Blade design, Bureau of Aeronautics 5868-9						
1	89	3	Single	Tractor	Radial-engine nacelle without wing	Biermann, David, and Hartman, Edwin P.: Tests of Five Full-Scale Pro- pellers in the Presence of a Radial and a Liquid- Cooled Engine Nacelle, Including Tests of Two Spinners. Rep. No. 642, NACA, 1938.
2	80	3	do	do	Liquid-tooled engine nacelle without wing	Biermann, David, and Hartman, Edwin P.: Tests of Two Full-Scale Pro- pellers with Different Pitch Distributions, at Blade Angles up to 60°. Rep. No. 656, NACA, 1939.
Blade design, Hamilton Standard 3155-6 and 3156-6						
3	90	2	Single	Tractor	Streamline nacelle without wing	Gray, W. H.: Wind- Trunnel Tests of Single- and Dual-Rotating Tractor Propellers at Low Blade Angles and of Two- and Three-Blade Tractor Pro- pellers at Blade Angles up to 65°. NACA A.R.R., Feb. 1943.
ţt.	90	3	do	do	do	Biermann, David, Hartman, Edwin P., and Pepper, Edward: Full-Scale Tests of Several Propellers Equipped with Spinners, Cuffs, Airfoil and Round Shanks, and NACA 16-Series Sections. NACA A.C.R., Oct. 1940.
5	90	3	do	do	Streamline nacelle with wing	Gray, W. H.: Wind- Tunnel Tests of Single- and Dual-Rotating Tractor Propellers at Low Blade Angles and of Two- and Three-Blade Tractor Pro- pellers at Blade Angles up to 65°. NACA A.R.R., Feb. 1943.
6	90	4	do	do	do	Biermann, David, and
7	90	4	do	do	do	Hartman, Edwin P.: Wind- Tunnel Tests of Four-
8	90	6	do	do	do	and Six-Blade Single- and Dual-Rotating Tractor Propellers. Rep. No. 747, NACA, 1942.
9	90	8	do	do	do	Biermann, David, and Gray, W. H.: Wind-Tunnel Tests of Eight-Blade Single- and Dual-Rotating Propellers in the Tractor Position. NACA A.R.R., Nov. 1941.
10	90	3	Single	Pusher	Streamline nacelle	Biermann, David, and Gray,
11 12 13	90 90 90	H 68	do Dual do	do do	without wing	W. H.: Wind-Tunnel Tests of Single- and Dual- Rotating Pusher Propellers Having from Two to Eight Blades. NACA A.R.R., Feb. 1942.
Blade design, Hamilton Standard 3155-6-1.5 and 3156-6-1.5						
ъ.	135	3	Single	Tractor	Streamline nacelle with wing	Biermann, David, Gray,
15 16 17	135 135 135	<b>6</b> 8	Dual	do do	do	W. H., and Maynard, Julian D.: Wind-Tunnel Tests of Single- and Dual- Rotating Tractor Propellers of Large Blade Width. NACA A.R.H., Sept. 1942.

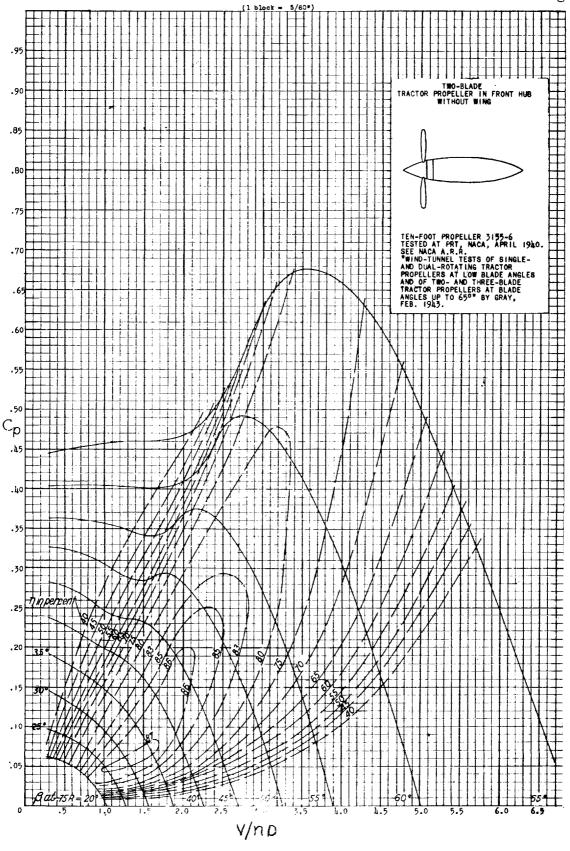






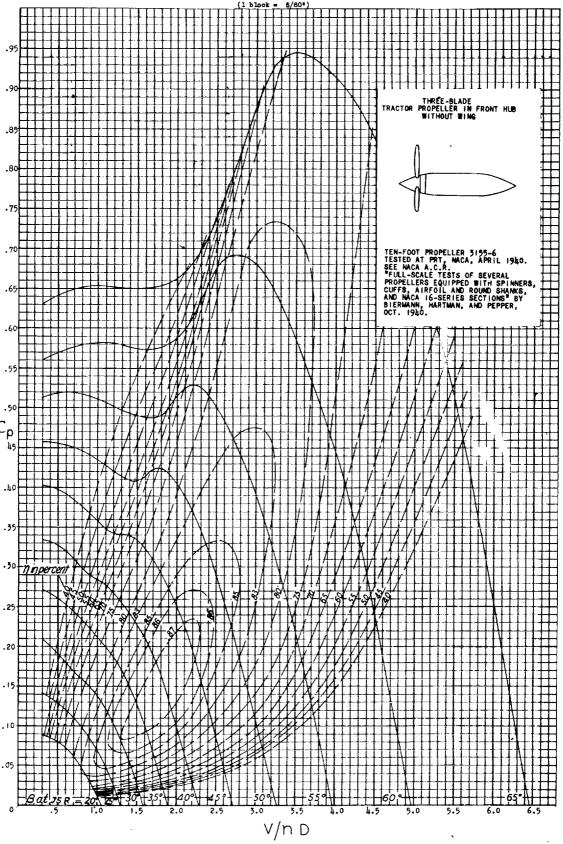
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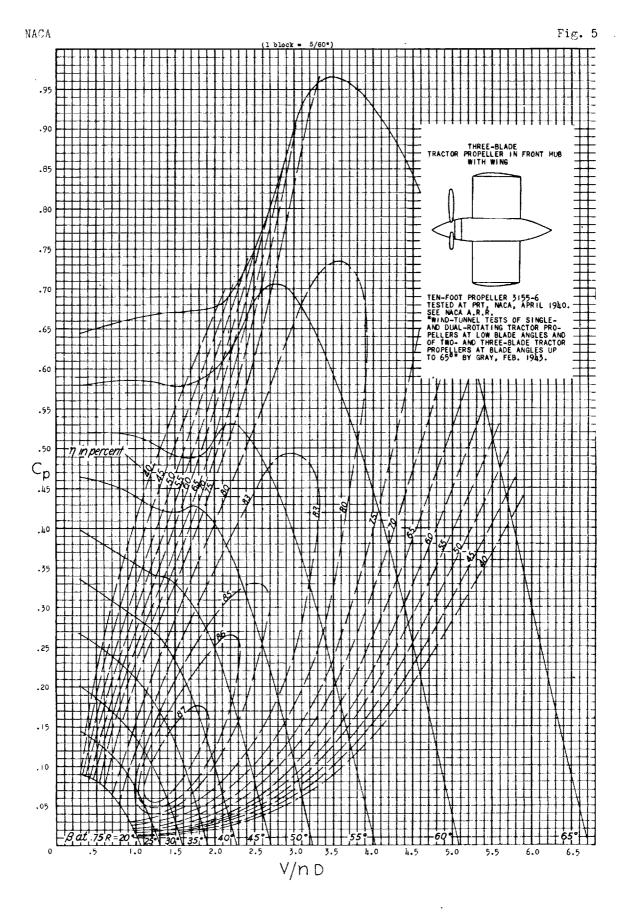
Fig. 3



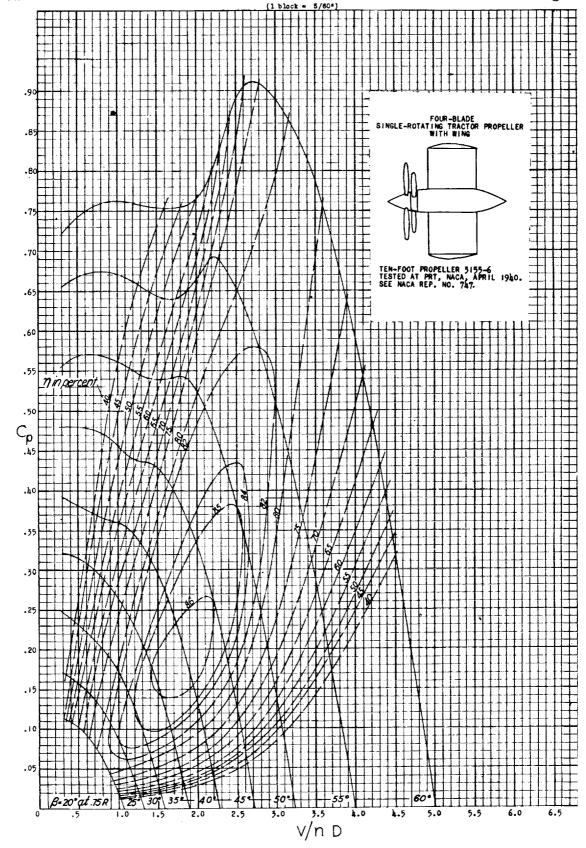
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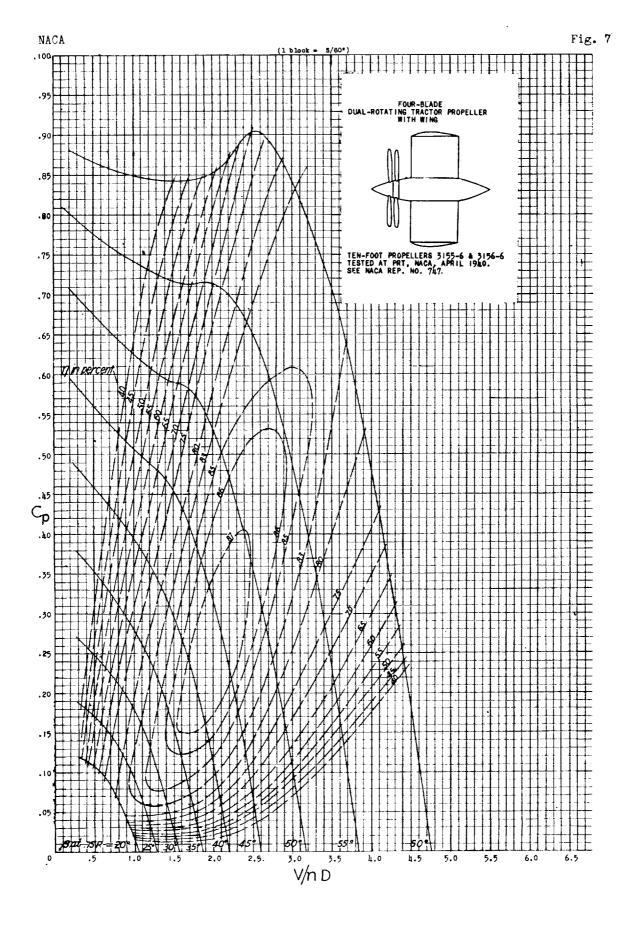
Fig. 4

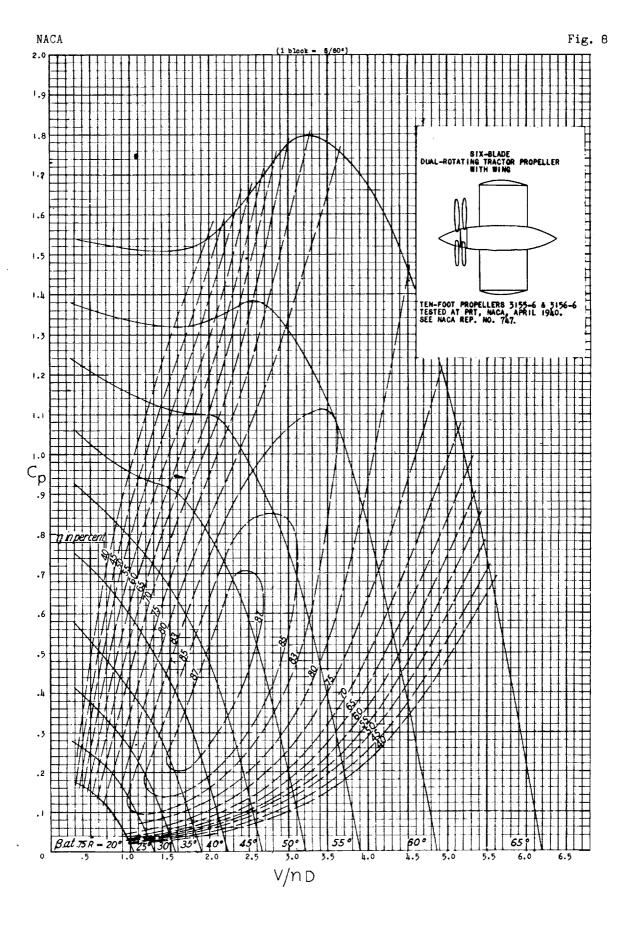




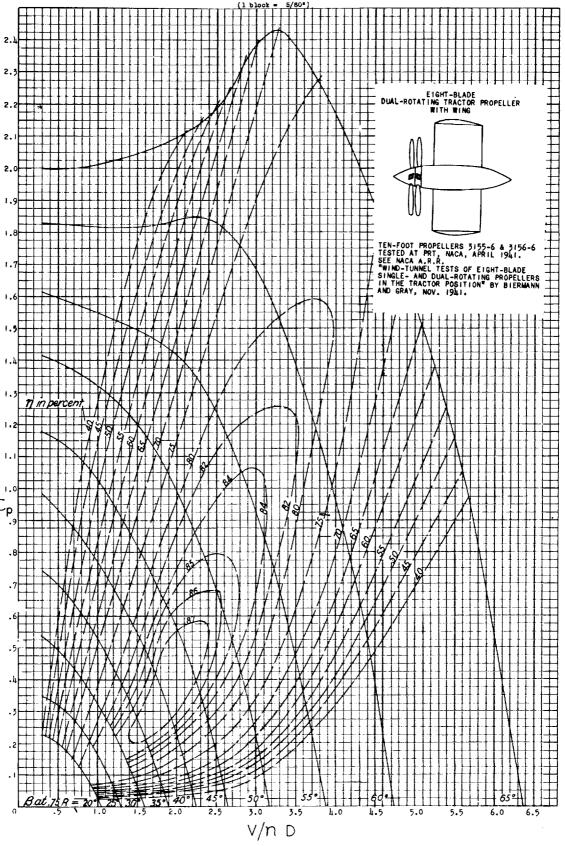
NACA Fig. 6

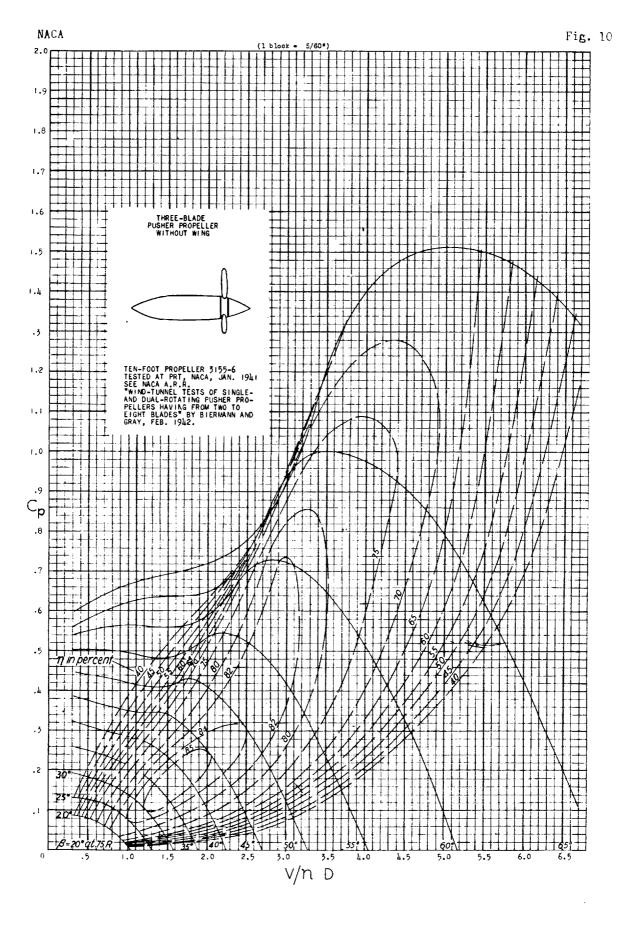




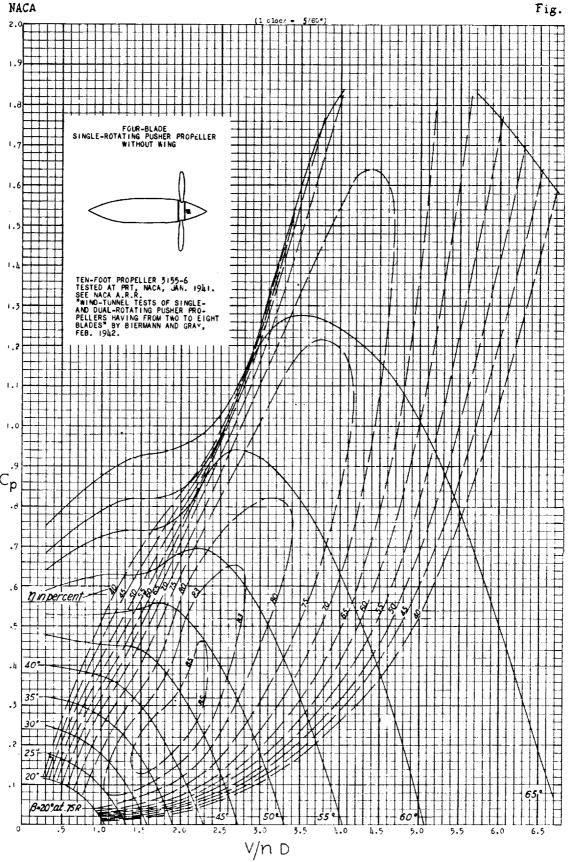


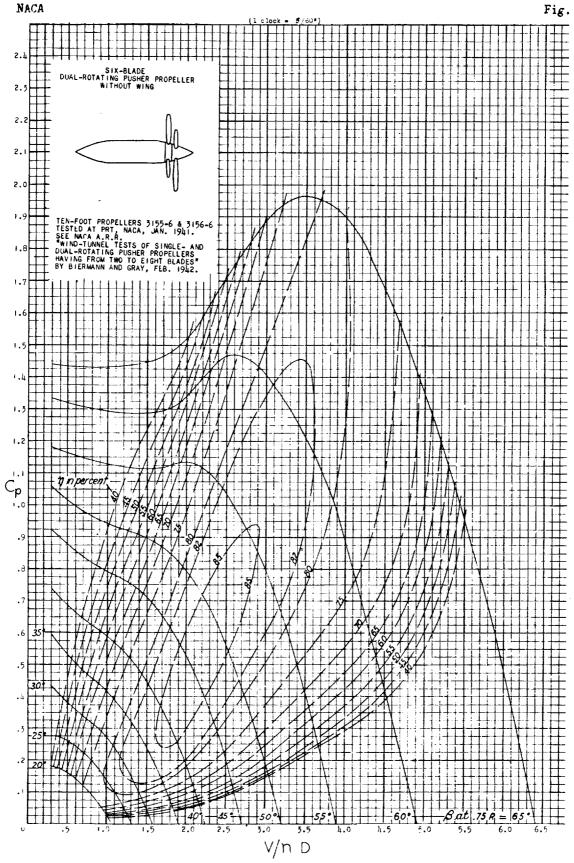
NACA Fig. 9

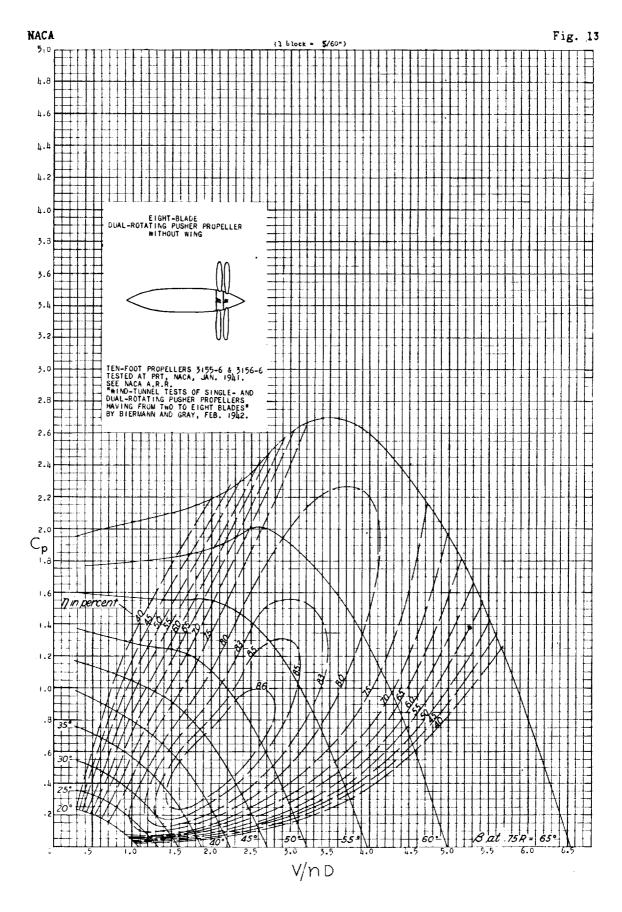


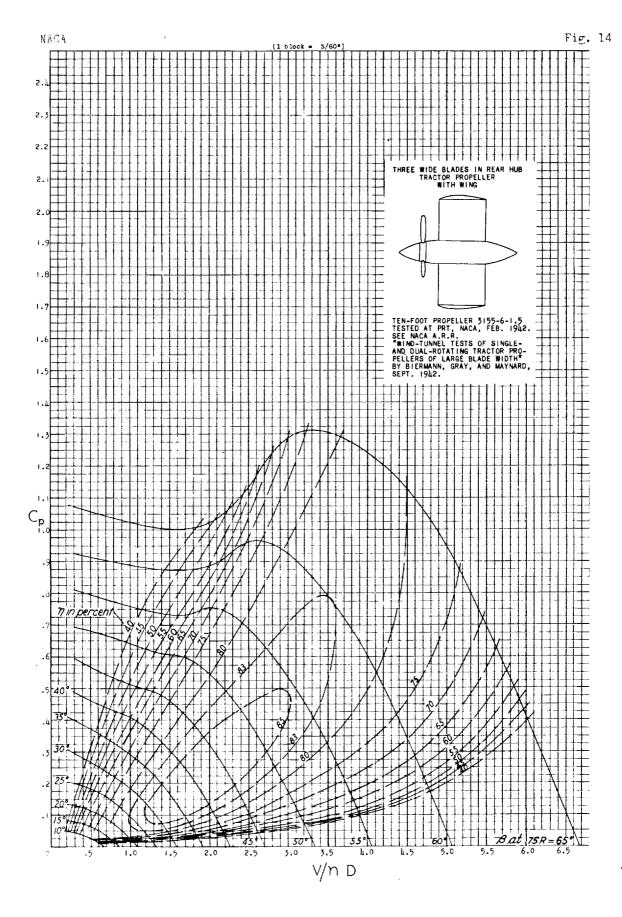


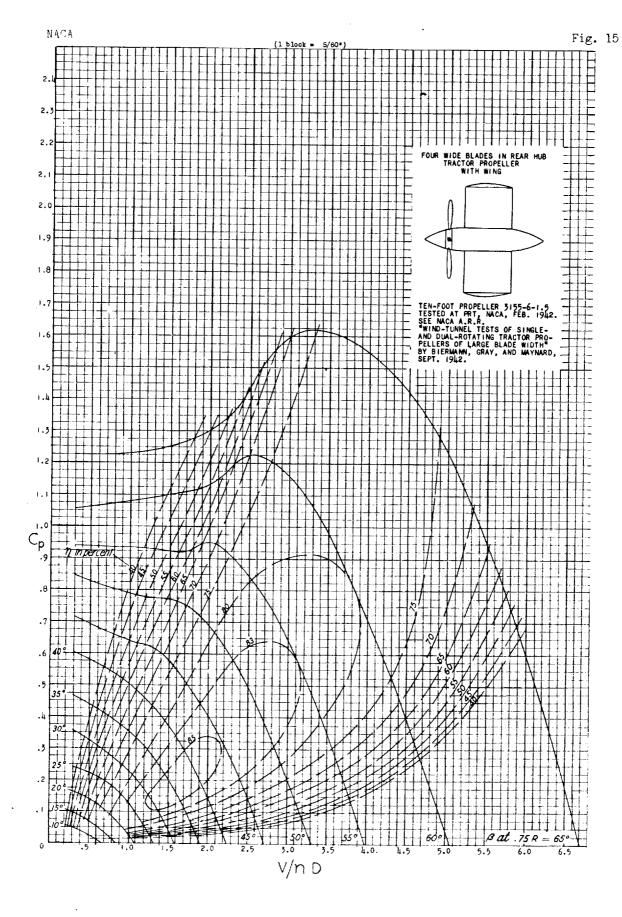












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